NEW CONCEPTS FOR LIGHT MECHANICAL STRUCTURES OF CYLINDRICAL DRIFT CHAMBERS



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ABSTRACT

A significant reduction in the amount of material at the end plates of a drift chamber can be obtained by the simple consideration of separating, in the mechanical structure, the gas containment function from the wire tension support function. According to this scheme, the wires are anchored to a self-sustaining light structure ("wire cage") surrounded by a very thin skin ("gas envelope") of suitable profile to compensate for the gas differential pressure with respect to the outside.

The "wire cage" is schematically made of a set of radial spokes, constrained into a polygonal shape at the inner ends and extended to the outer endplate rim, thus subdividing the chamber in identical sectors. The drift chamber is, then, built by stacking up radially, in each of the sectors and between adjacent spokes, printed circuit boards, where the ends of the wires are soldered, alternated with proper spacers, to define the cell width.

A system of adjustable tie-rods steers the wire tension to the outer endplate rim, where a rigid cylindrical carbon fibre support structure, bearing the total wire load, is attached.

Two thin carbon fibre domes, free to deform under the gas pressure without affecting the wire tension and conveniently shaped to minimize the stress at the inner rim, contribute to the "gas envelope" and, together with an inner thin cylindrical foil and with the outer structural support, enclose the gas volume.



Each **spoke** is under tension against the structural outer cylinder shell. This tension is transferred to the **spoke retainers**, which are all constrained by their polygonal structure.

Stereo angle development

The distribution on the spokes of the load due to the wire tension per unit length is a function of the number of layers (36 here), the number of cells per sector (16) and the number of sectors (24). Assuming 1 cm² average size cells and 150 g (5 field wires per sense wire) tension per cell, each spoke feels a flexural stress of 60 Kg (1.5 Kg/cm) which must be compensated by the tension applied to the **stays**. Assuming one stay every 4 layers, 9 stays total, the tension to be applied to each stay amounts to less than 20 Kg. This amounts to a maximum local transverse deviation along each spoke of less than 10 µm.



GAS ENVELOPE

End-plate profile optimization by **multivariate analysis**

- Use isotropic material (1 mm thick Aluminum) solid rotational plates + inner cylinder (ideal joints)
- > Assume infinitely rigid outer cylinder
- Parameterize geometry by:
 - constraining inner cylinder radius
 - constraining inner cylinder length
 - constraining outer cylinder radius
 - varying middle point of a 3 point-spline profile for the end plates
- > Optimize dynamic properties:
 - minimum stress at inner boundary
 - minimum of maximum stress
 - maximum safety factor
- No convergence for isotropic material
- Replace isotropic material with light composite material





parameters	Initial model	Optimized model
Maximum stress	357.5 MPa	58.7 MPa
Stress at inner boundary	267.4 MPa	26.6 MPa
Safety factor	0.783	4.44

Replace isotropic material with composite material

- Chose proper unidirectional prepreg and form ply
- Check for feasibility of the chosen geometry
- Draping of the laminates and flat-wrap of the model (align the fibers along the direction of stress)
- Static structural analysis for different configurations and different choices of prepreg



stress

Static solution with 4-ply Orthotropic (0-90-90-0) HM M30S 53 ET443 51% 60 μm/ply - 0.0053 g/cm² but Inner cylinder unstable for buckling

unless ...



failure crite

Solve buckling instabilities

Case of isotropic 1 mm Al:

critical load / applied load = load multiplier = 0.092 !

Critical load proportional to cylinder moment of inertia proportional to thickness⁴

Solution is sandwich of 2 C-fiber skins, 2-ply each, HM M30S 53 ET443 51% with a 4.76 mm C-foam core **Grafoam® FPA-10** for a total areal mass of 0.100 g/cm² total



Inner cylinder sandwich: 2 C-fiber skins, 2-ply each, HM M30S 53 ET443 51% + C-foam core, 5 mm total **Grafoam® FPA-10** 0.100 g/cm² total

> End plates: 4-ply orthotropic (0-90-90-0)

WIRE CAGE MATERIAL BUDGET GAS ENVELOPE											
	[g/cm ²]	X ₀ [10 ⁻³]	Hit probability	[g/cm²]	X ₀ [10 ⁻³]		type	# of ply - thick	[g/cm²]	X ₀ [10 ⁻³]	[Kg]
Spokes	0.397	9.30	0.0225	0.009	0.2	Innor	HM M30S 53 ET443	2 ply – 120 µm	0.011		
Stays	0.009	0.19	0.00775	0.009	0.2	Inner Cylinder	Grafoam® FPA-10 HM M30S 53 ET443	4.76 mm 2 ply – 120 μm	0.079	1.95	4.50
Spokes connecting ring	0.850	19.9	0.023	0.019	0.5				0.011		
Spacers	0.074	1.80	0.878	0.065	1.6	Inner ring	C-fiber inserts (2.3%)	10 mm 232.5 cm ²	1.700 0.040	39.81	0.10
field wires board	0.720	2.64	0.05	0.036	1.3		(2.570)	60 cm ³		0.92	
sense wire board	1.800	6.59	0.024	0.045	1.6	End plates	HM M30S 53 ET443	4 ply – 240 µm	0.021	0.41	0.26
R/O board components				0.027	1.2	Outer	C fiber incorto				
HV board components				0.047	2.0	ring	C-fiber inserts				
R/O cables				0.001	0.0	Outer	 Grafoam® FPA-20	32 ply – 2.5 mm 10 mm	0.325 0.324		
HV cables				0.019	1.2	Cylinder		32 ply – 2.5 mm	0.325		

		Wire Cage		Gas Envelope		TOTAL	
		[g/cm ²]	X ₀ [10 ⁻³]	[g/cm ²]	X ₀ [10 ⁻³]	[g/cm ²]	X ₀ [10 ⁻³]
	Spokes + Stays	0.009	0.2	0.021	0.4	0.203	6.3
Upstream End-Plate	Spacers	0.065	1.6				
	Wires board	0.081	2.9				
	R/O components	0.027	1.2				
Downstream	Total above	0.182	5.9				
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Work originally developed for the **I-tracker** drift chamber proposed for the Mu2e experiment at Fermilab